

AURORA

Aerocapture **U**ranus p**R**obe & **O**rbiter for **R**econnaissance & **A**nalysis

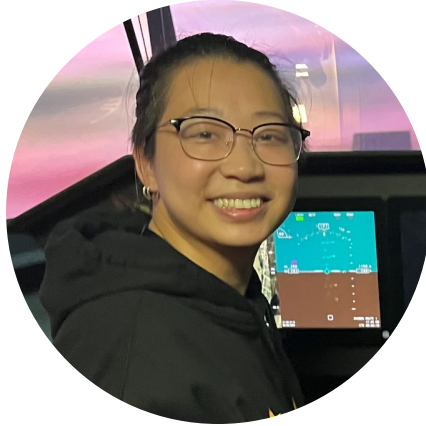
Team 8 - Julianne,
Jace, Tamara, Cynthia

Introductions - Team 8



Julianne Miana

*Mission Ops & Scientific
Instrumentation*



Jace Lu

*Mission Ops &
Trajectory Planning*



Cynthia Cao

*Scientific & Technical
Motivation*



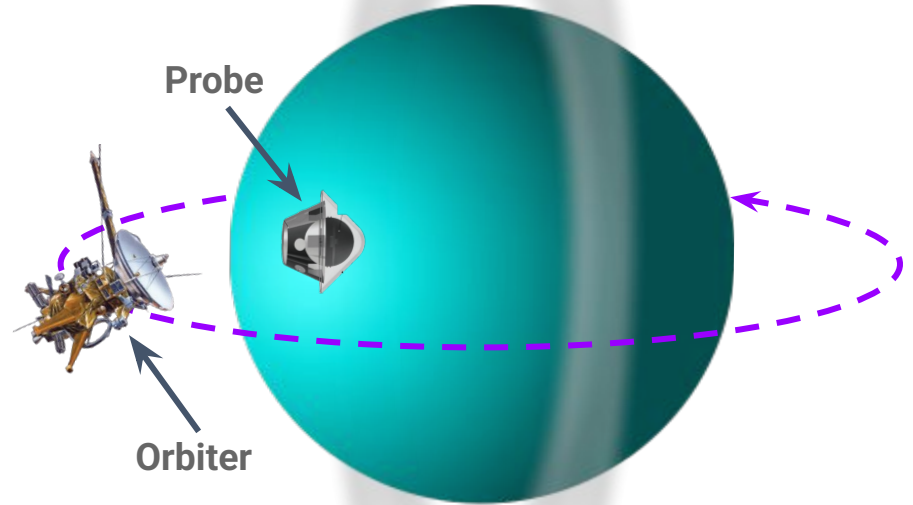
Tamara Hinderman

*Pitch Team Lead &
Technical Motivation*

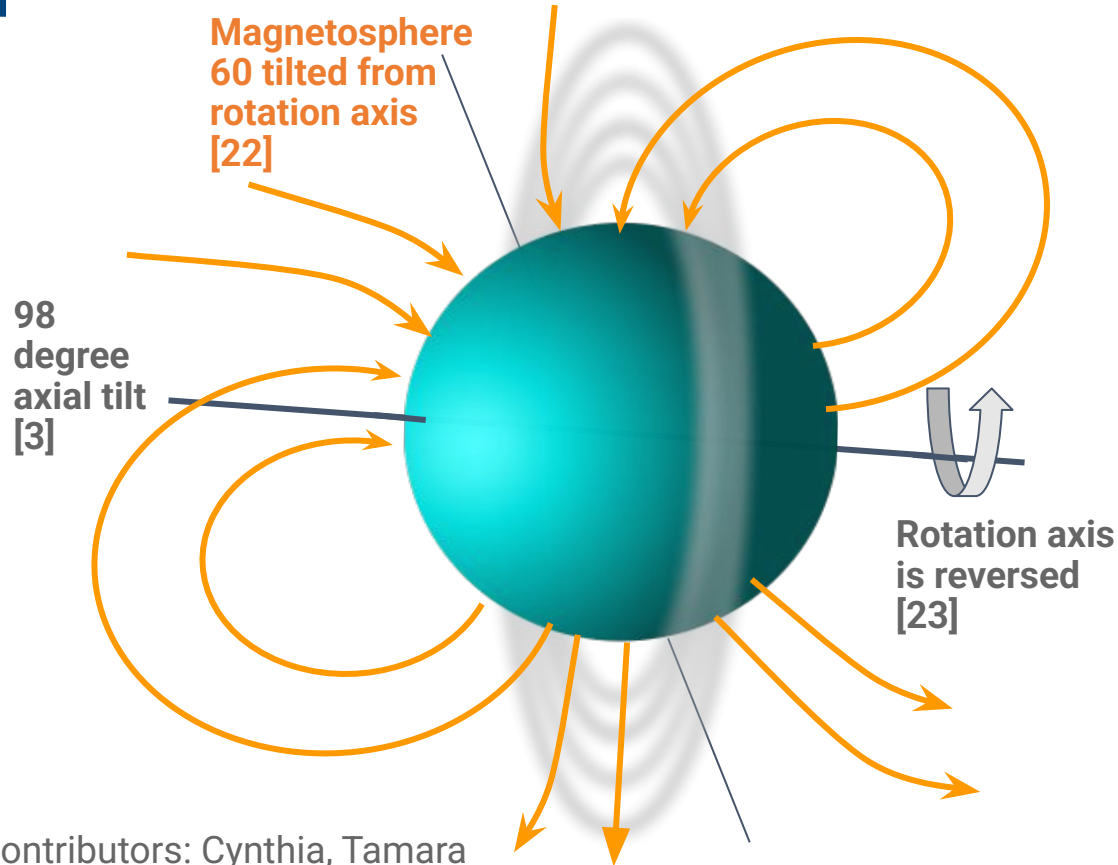
Revisiting Uranus for the first time since Voyager 2

AURORA:

- Sending an orbiter and probe to Uranus to investigate
 - planet's properties
 - atmospheric conditions
 - exoplanet origins
- Last time: Voyager 2 flyby (1986)
- 1st orbiter & probe reconnaissance mission



Researching Planetary Formation Through Tilt & Magnetosphere



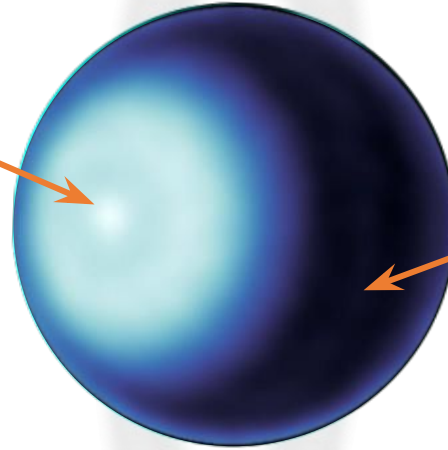
Investigating Uranus' Extreme Atmospheric Conditions

Coldest atmosphere in
solar system:

-320 F

-196 C

Little understanding of heat
flux and internal heat [14]



Unclear atmospheric
circulation from interior
to thermosphere [14]

Leveraging Uranus to Explore Exoplanet Origins



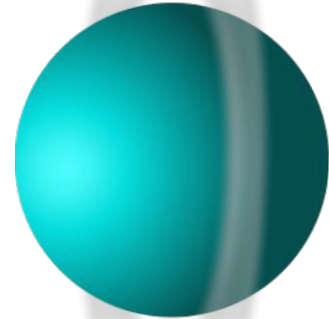
TOI-421 b
(Hot Sub-Neptune)



GJ 1214 b
(Warm Sub-Neptune)

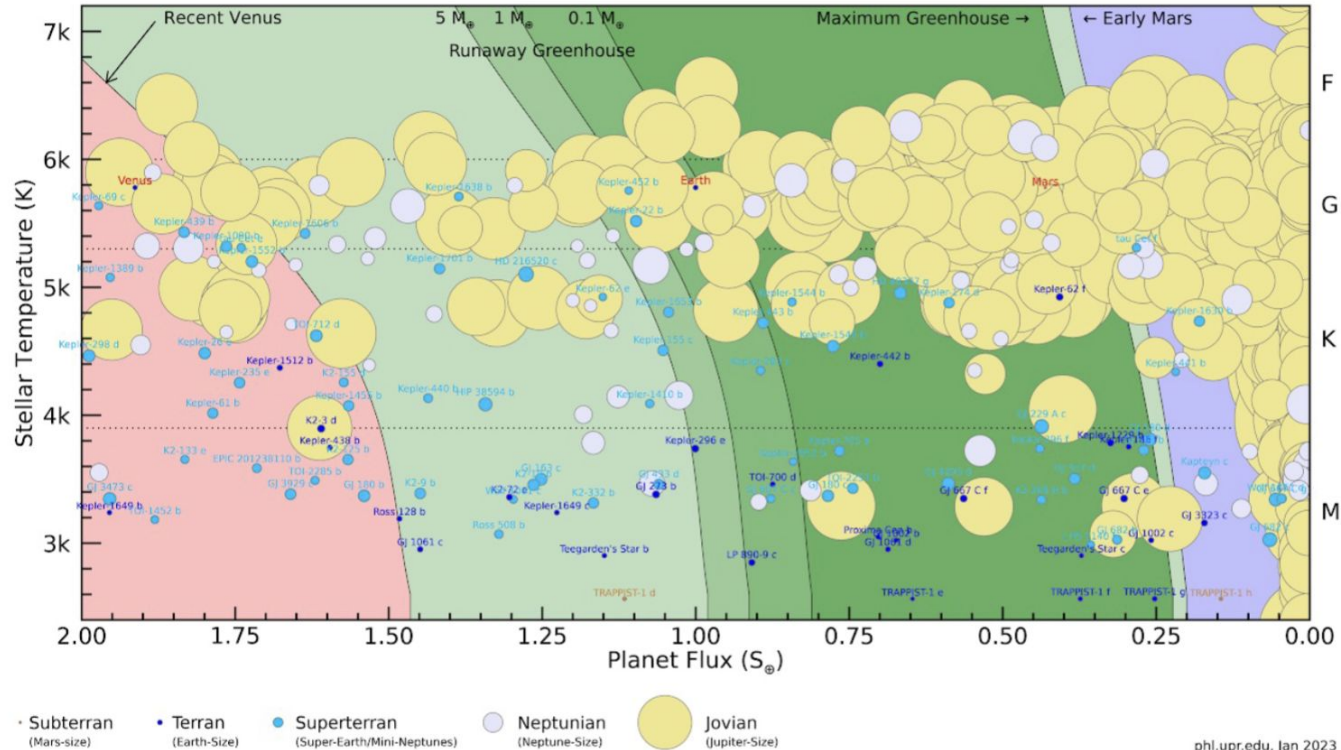


Neptune
(Ice Giant)



Uranus
(Ice Giant)

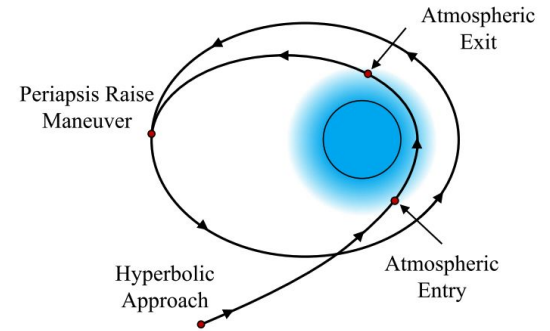
Leveraging Uranus to Explore Exoplanet Origins



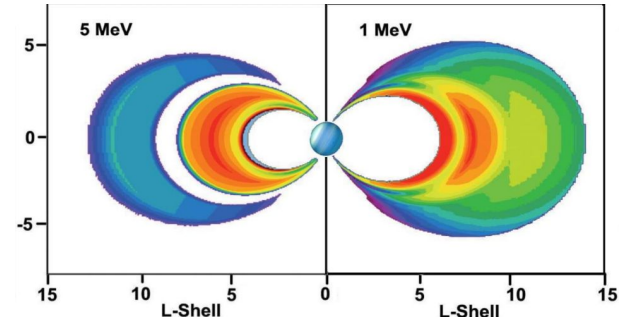
Why AURORA?

Testing novel concepts for novel problems

- Using aerocapture for the first time at an outer planet to decelerate into stable orbit [1][9][12][16]
- Designing spacecraft to handle extreme conditions
 - Orbit/tilt around Uranus [23]
 - Magnetic field + radiation [11][22]
 - Extreme temperatures + winds [14][16]

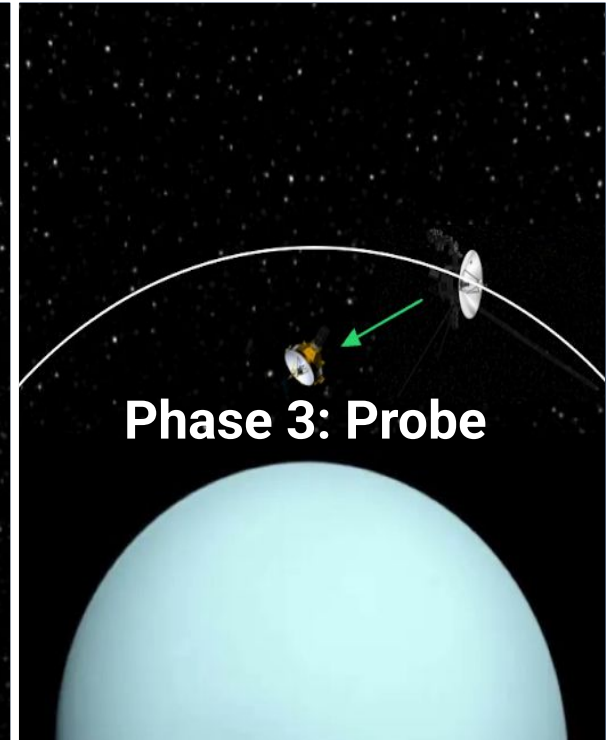
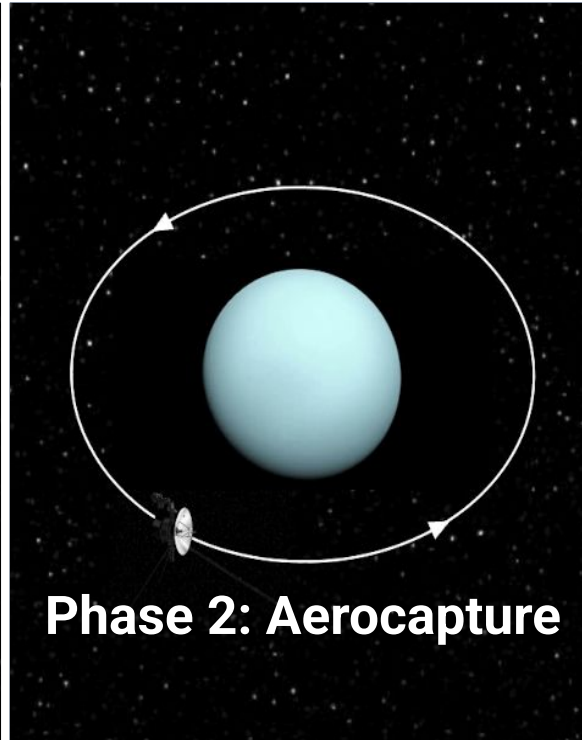
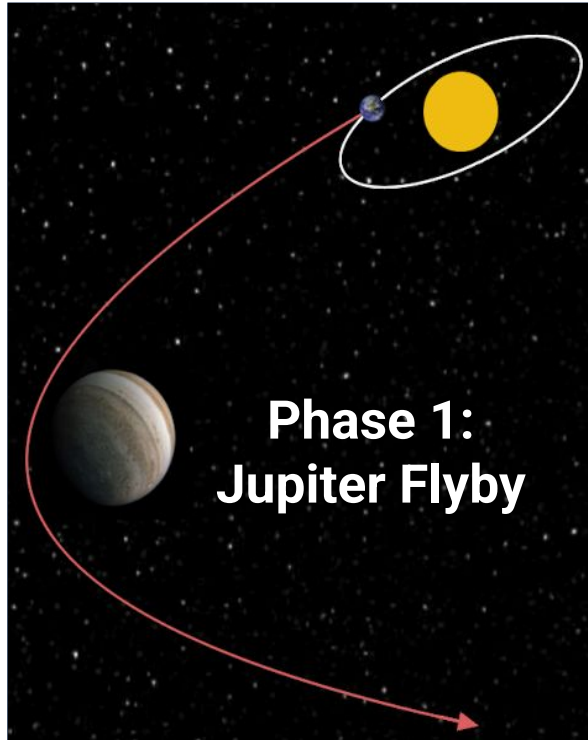


Aerocapture Diagram [12]



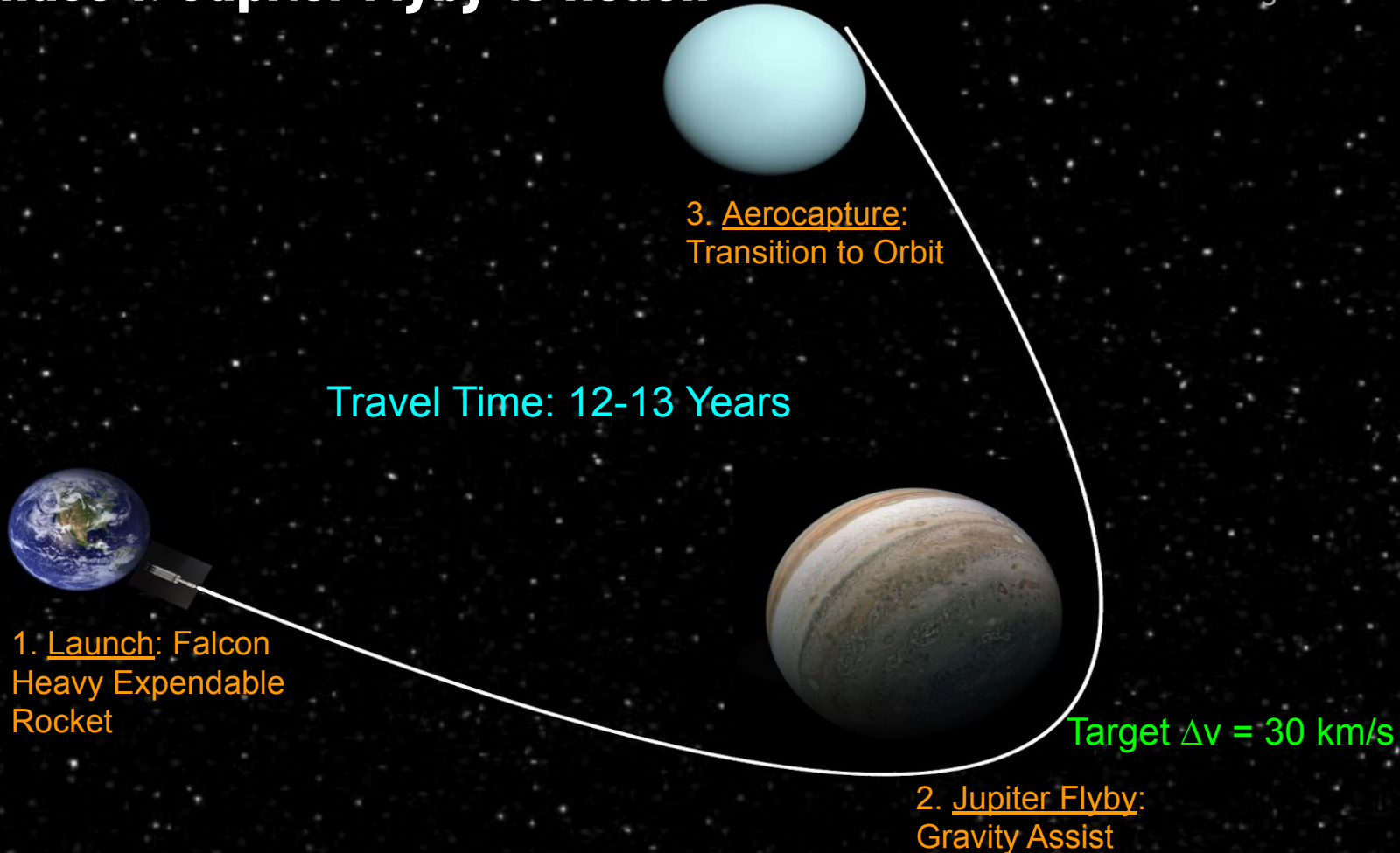
Uranian Radiation Belt Model [11]

Conops



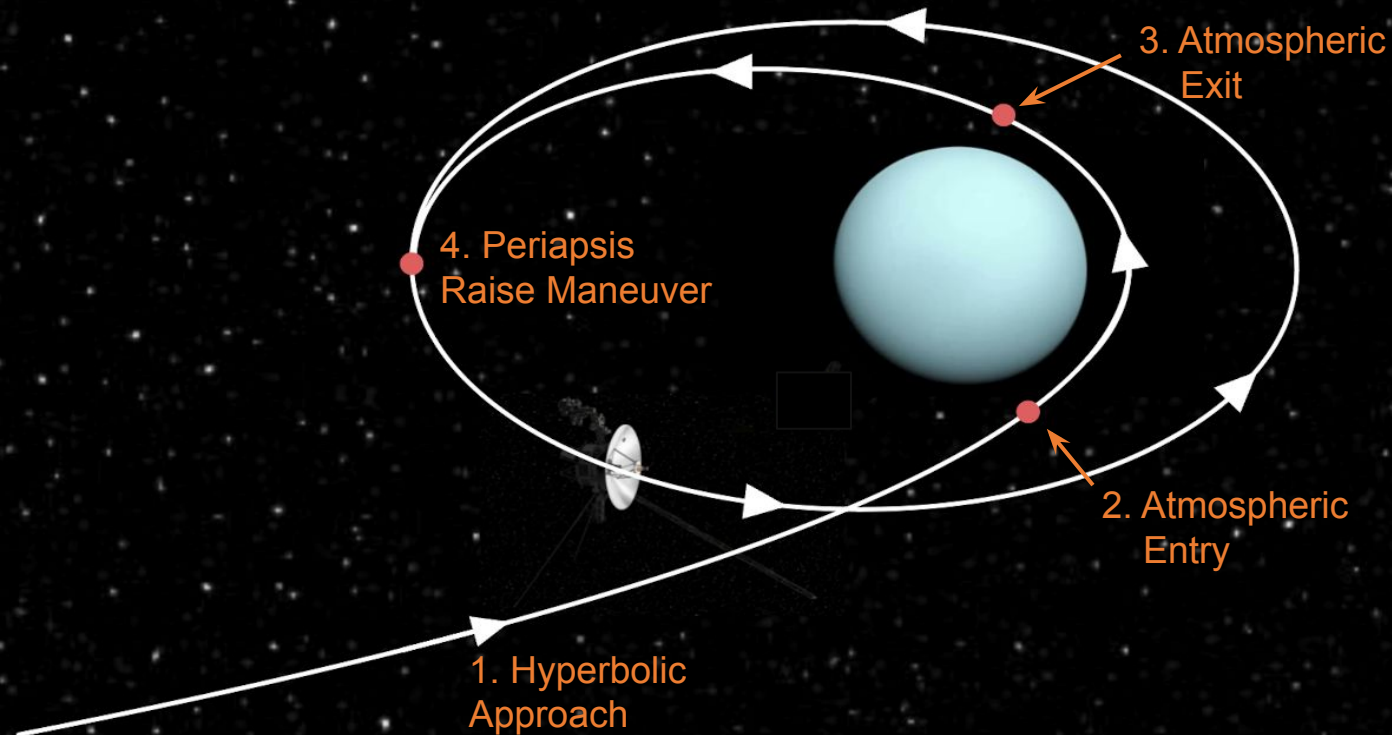
CONOPS Phase 1: Jupiter Flyby to Reach Uranus

Contributors: Jace
Presenting: Jace



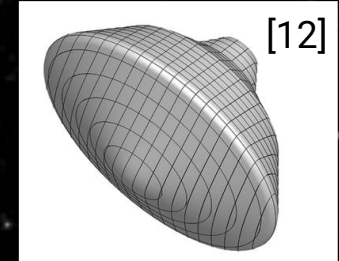
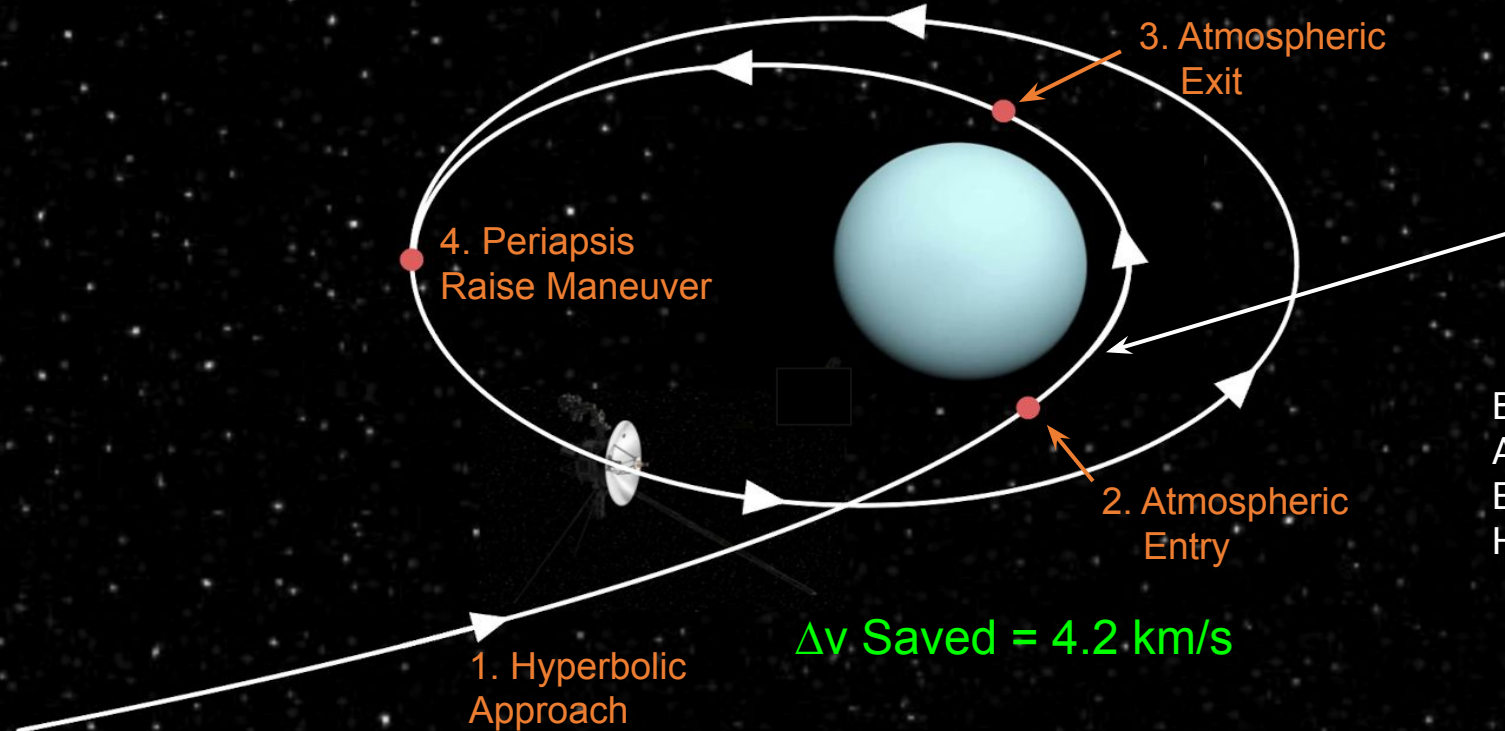
CONOPS Phase 2: Aerocapture to Transition to Orbit

Contributors: Jace
Presenting: Jace



CONOPS Phase 2: Aerocapture to Transition to Orbit

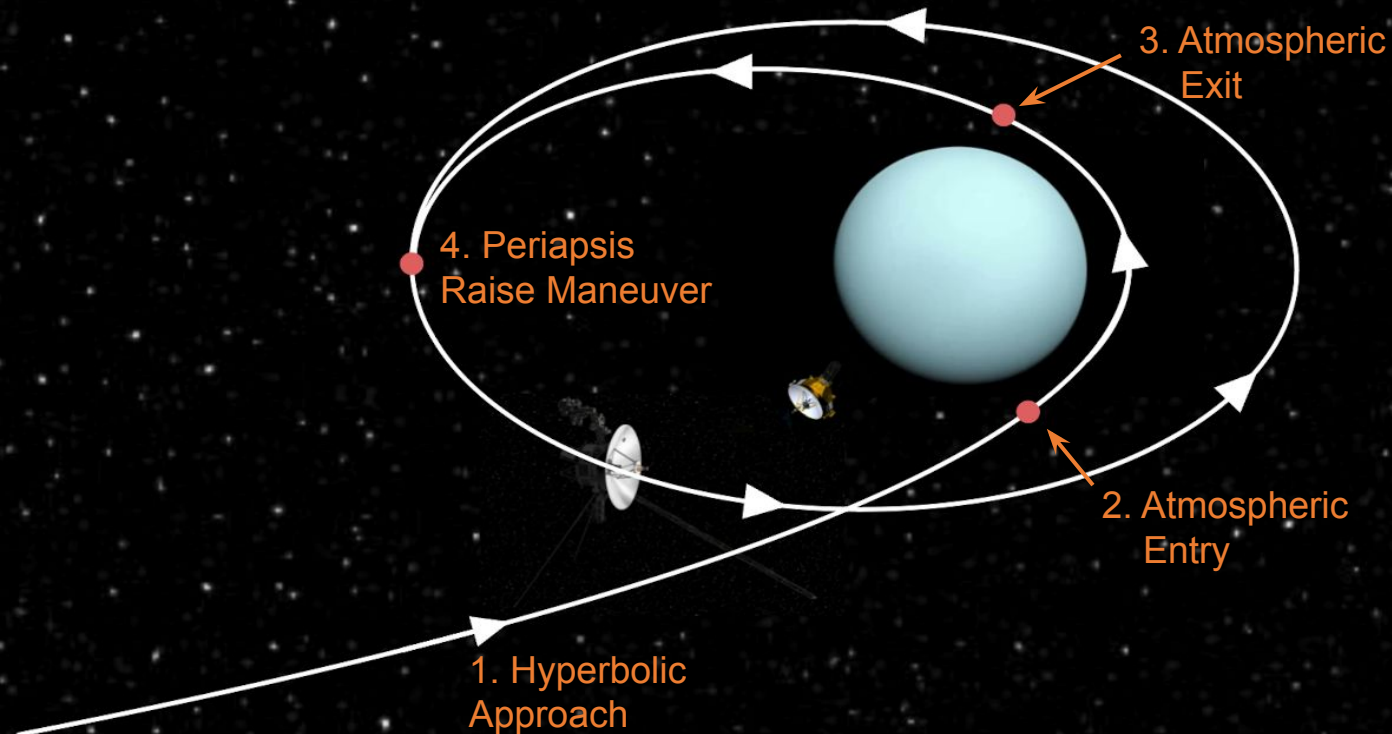
Contributors: Jace
Presenting: Jace



Blunt Body: $L/D = 0.25$
Altitude: 1000km
Entry Velocity: 23.78km/s
HEET/other TPS Material

CONOPS Phase 3: Launching Probe to Collect Data

Contributors: Jace
Presenting: Jace



CONOPS Phase 3: Launching Probe to Collect Data

Contributors: Jace
Presenting: Julianne

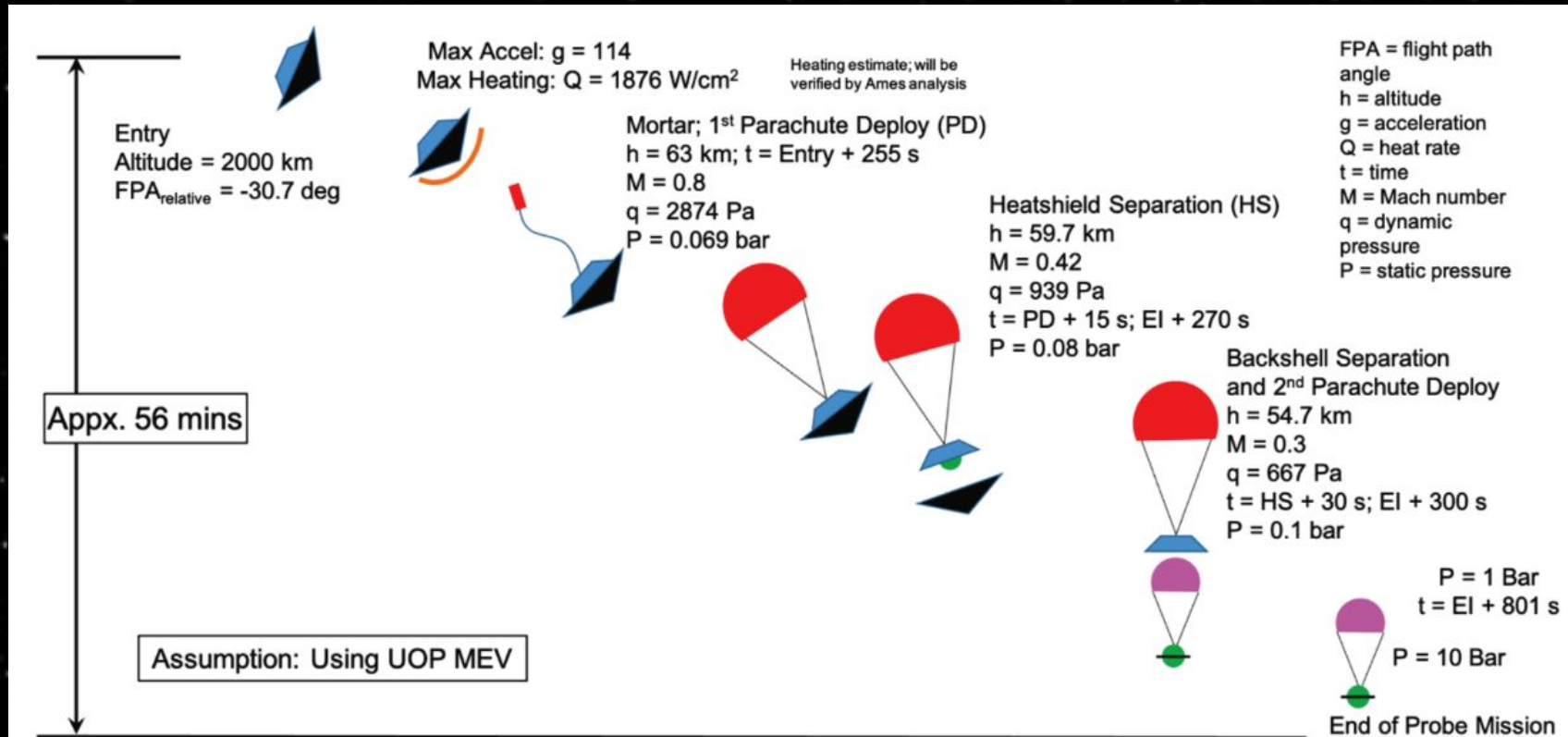
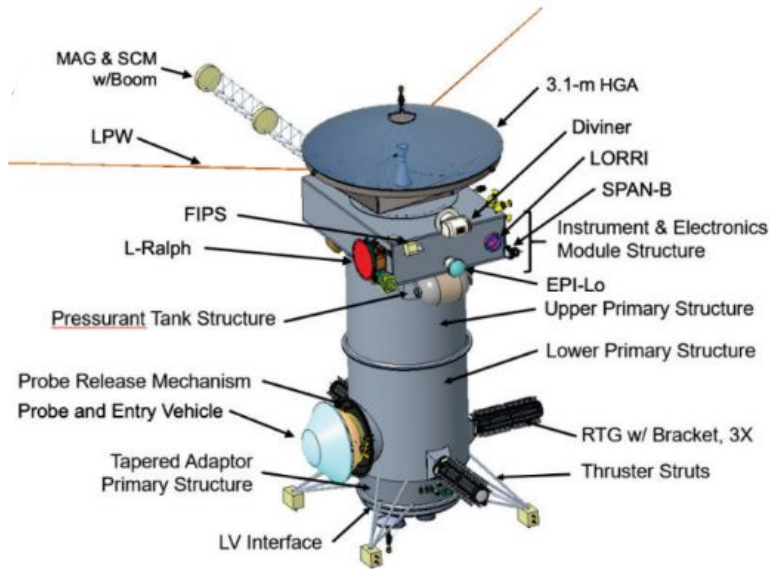


Exhibit 3-34. Probe Trajectory Concept of Operations.

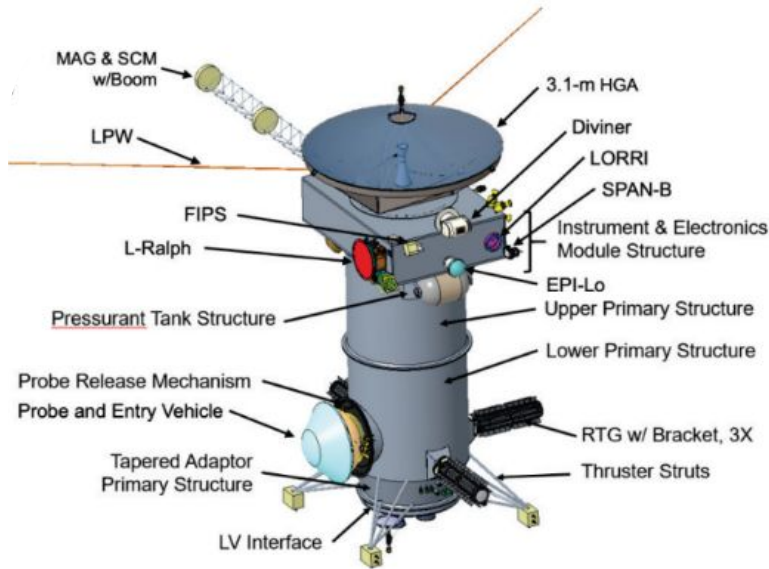
Utilizing existing orbiter technology



2021 NASA Uranus Orbiter & Probe Mission
Concept Study [7]

- Imaging [7]
 - Thermal IR Camera: **characterize thermal environment** (LRO Diviner)
 - Magnetometer + Boom: **measure magnetic field** (MESSENGER, Cassini)
- Scientific analysis [7]
 - Fields & particles package:
 - EPI-Lo: **characterize energetic particles** (Solar Parker Probe)

Utilizing existing orbiter technology



2021 NASA Uranus Orbiter & Probe Mission
Concept Study [7]

- Thermal / Material [7][10]
 - HEEET TPS
 - Anodized Al shielding
- Comms [7]
 - Radio Science Experiment (MESSENGER / RS)
- Power
 - RTGs

Combining existing and proposed probe technologies

- Scientific Payload: [7]
 - Atmospheric Structure Instrument: **measure properties of atmosphere** (Cassini-Huygens)
 - Mass Spectrometer: **identify unknown compounds** (Rosetta DFMS)
 - Ortho-Para Instrument: **detect and measure hydrogen isomer ratio**
- Thermal
 - Single-layer HEEET TPS
- Power
 - 2 lithium batteries

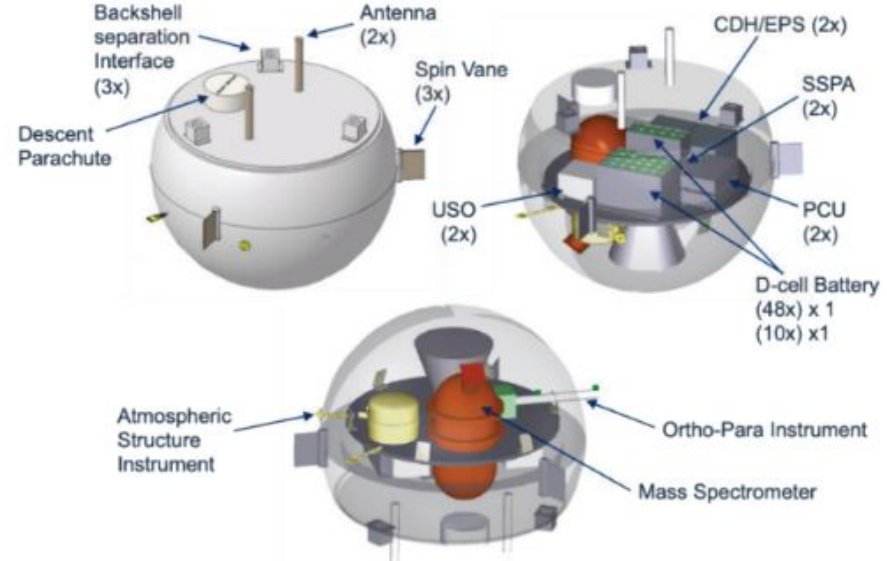
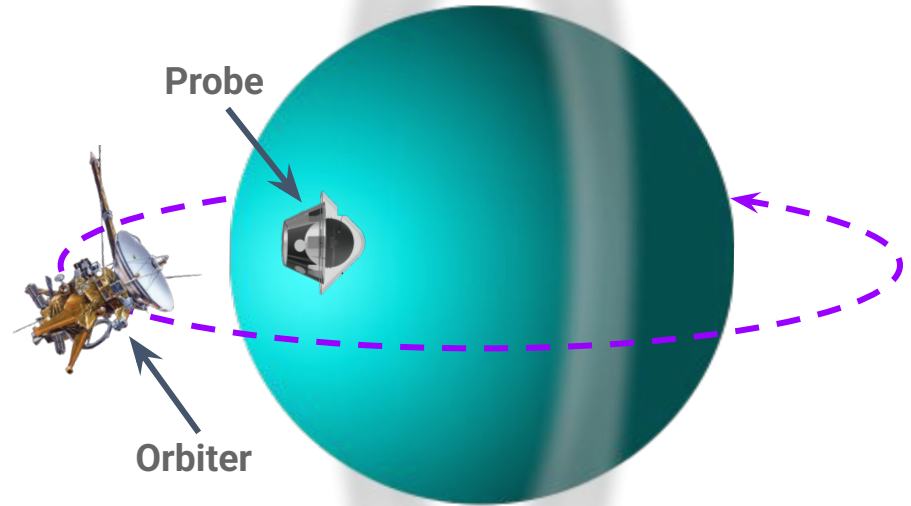


Exhibit 3-19. Descent module configuration.

2021 NASA Uranus Orbiter & Probe Mission Concept Study [7]

AURORA Mission Value

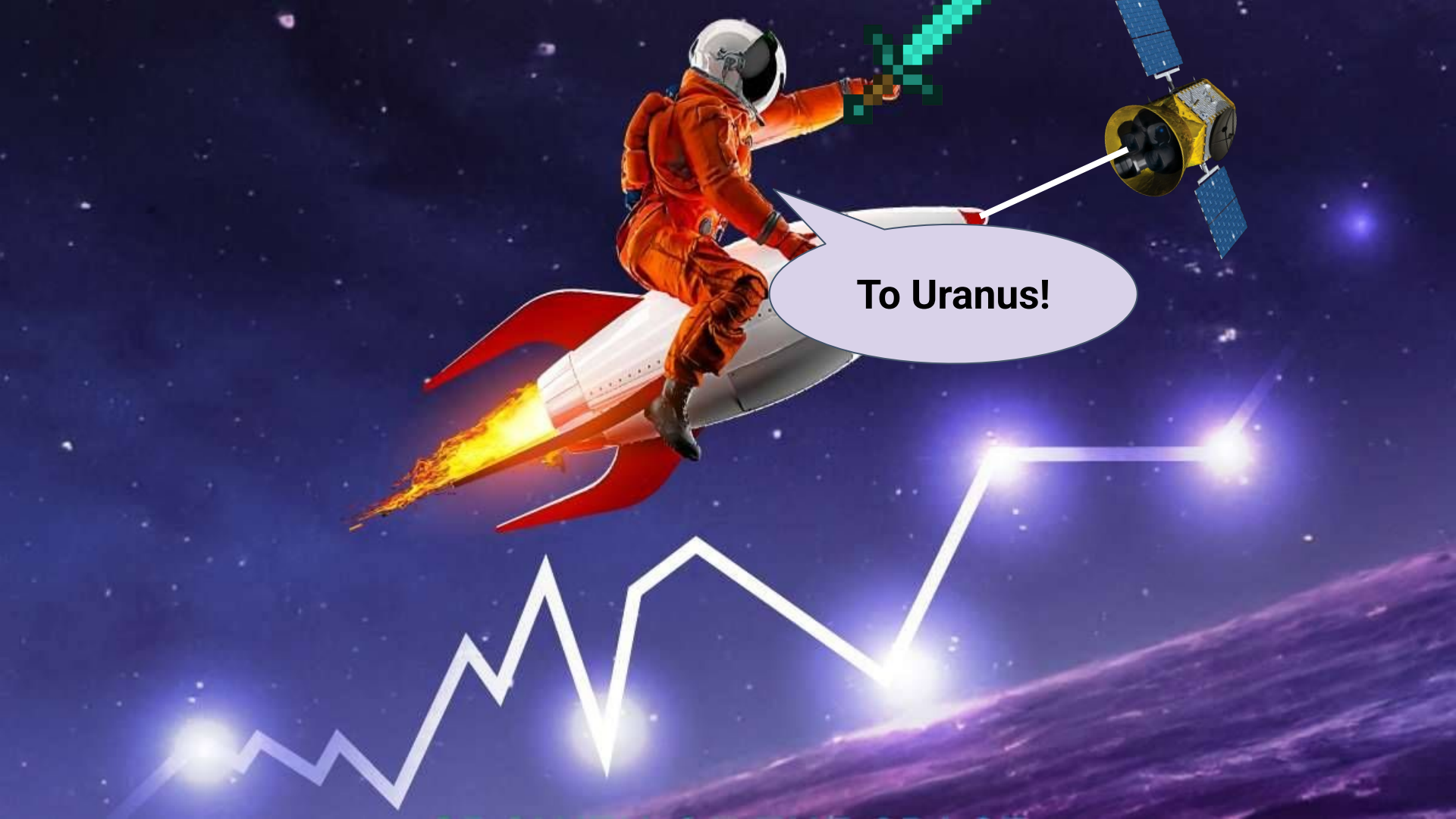
- First orbiter and probe for investigating Uranus
- Fulfills multiple scientifically valuable objectives
- Challenging design environment
- Innovating on flight heritage technology to solve a novel problem



AURORA

Together we can explore the depths of Uranus!





To Uranus!

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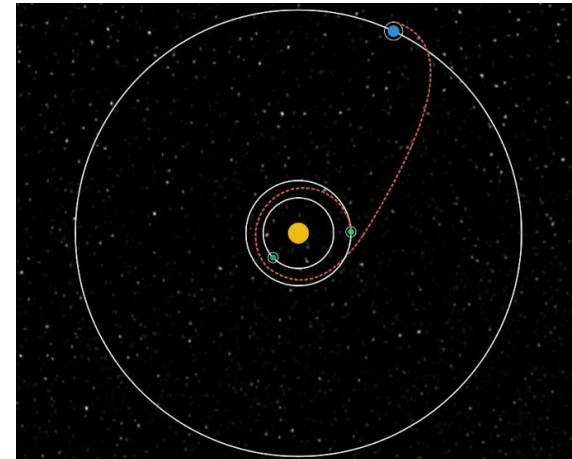
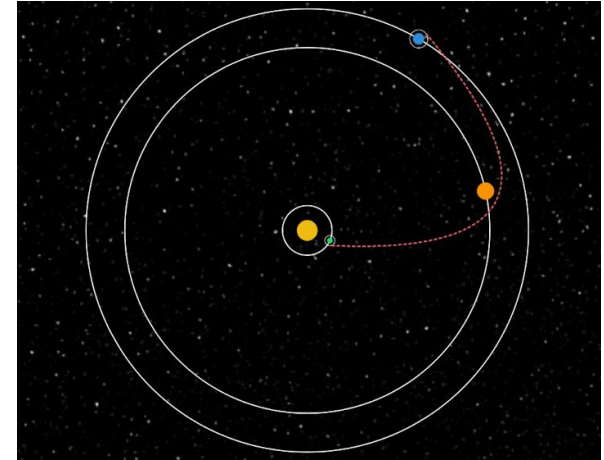
Bibliography

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Backup Slides

Reaching Uranus

1. Launch Window: 2031 - 2032
 - Launch Vehicle: Falcon heavy rocket [13]
 - Use Jupiter as a flyby [7]
 - Use Aerocapture to transition from a hyperbolic trajectory to elliptical orbit [12]
 - Arrival: 2044
2. Launch Window: After 2032 [7]
 - Launch Vehicle: Starship [13]
 - Use Earth and/or Venus as flybys
 - Flight Time: 14+ years



AURORA by the Numbers [7]

1. Mission Lifetime: 18.5 yrs
2. Total ΔV : 2708 m/s
 - a. Aerocapture dissipates up to a ΔV equivalent of 4.2 km/s
3. Mass Budget: 6552 kg total
 - a. Dry mass: 3429 kg
 - b. Wet mass: 3123 kg
4. Design Life
 - a. Orbiter: 18.5 yrs
 - b. Probe: 13.7 yrs
5. Cost: \$2.8B* (FY\$25)

Extra Math stuffs

<https://drive.google.com/drive/folders/1Qlun6EF0v472eOMXXokHxa6B2tBLNkdv> (NASA Mission Study on Uranus Orbiter + Probe Requested by 2023-2032 Decadal Survey)

https://discovery.larc.nasa.gov/PDF_FILES/Aerocapture_std_briefing_June10_v2.pdf

https://www.hou.usra.edu/meetings/icegiants2014/presentations/Da_y_3/Agrawal_2008.pdf - 3000 km

<https://arc.aiaa.org/doi/pdfplus/10.2514/6.2025-1510> - altitude 1000 km

Launch Feasibility with Starship

Contributors: Tamara
Presenting: Tamara

- Starship capable of direct transfer to Uranus [13]
- Starship UOI would require a larger propellant mass, most likely located in the main tanks with a high boil-off risk.

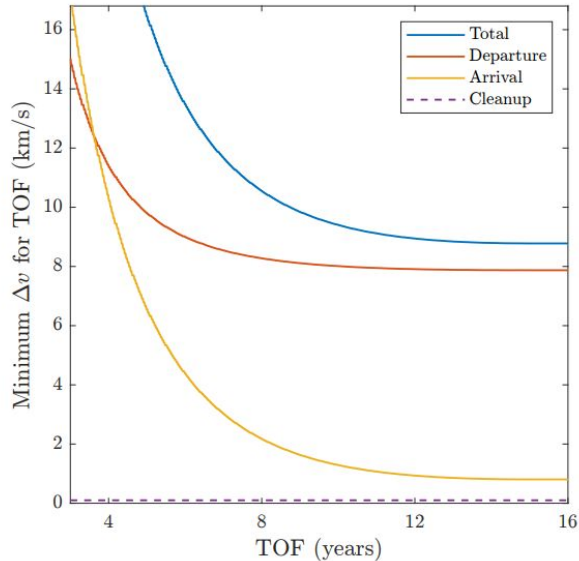


Figure 9: Plot of Δv and TOF tradeoff for Uranus direct transfer.

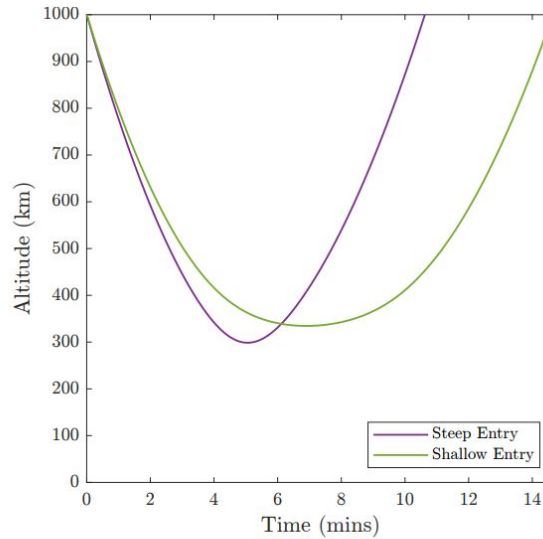


Figure 12: Example aerocapture bounding trajectories for Starship configuration with both fin sets deflected arriving with v_∞ of 8 km/s.

Table 7: Baseline UOP mass and propulsive properties, from [2].

Parameter	Value
Flight system launch mass	7235 kg
Inserted mission mass at Uranus	4919 kg
Total notional mission Δv	2708 m/s
Δv for cruise and UOI	1887 m/s
Δv available after UOI	821 m/s

Entering Uranus Orbit

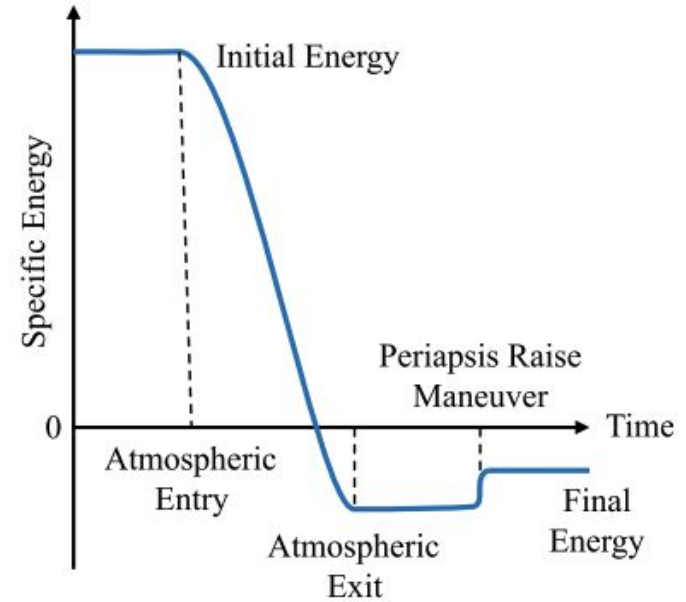
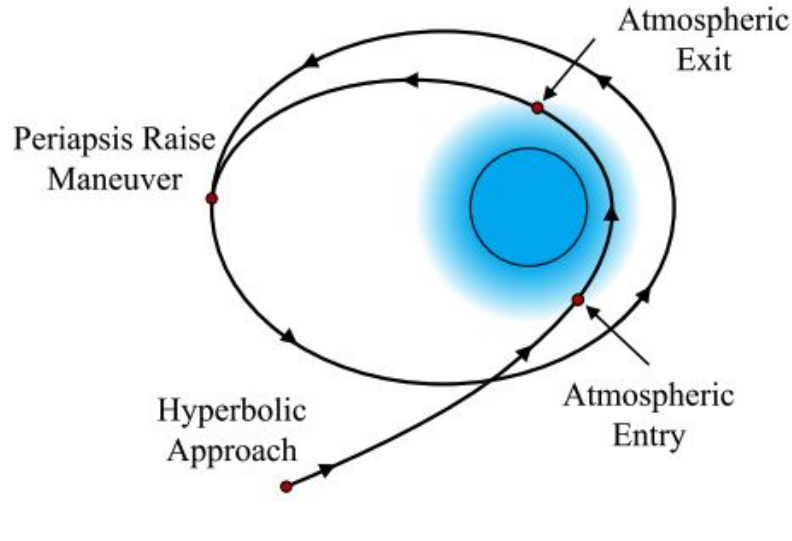
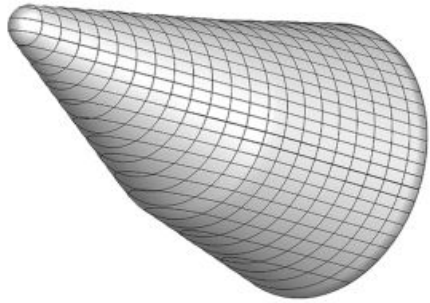
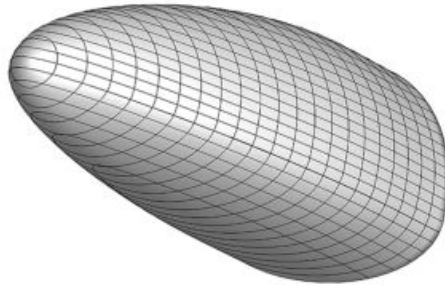


Fig. 1 Diagrams showing example aerocapture trajectory and spacecraft specific energy over time. [12]

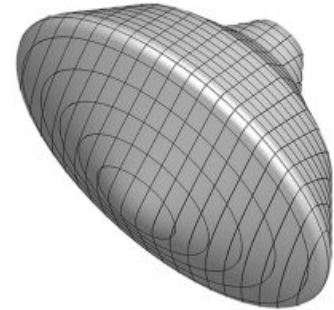
Heat Shield Shapes



a) Florence (1981), $L/D = 2.0$ [8]



b) Ellipsled (2004), $L/D = 0.6$ [10]



c) Blunt-Body (2023), $L/D = 0.25$ [7]

[12]

Mission Δv

Phase	Maneuver Name	Δv (m/s) Deterministic	Δv (m/s) Statistical
Cruise	Launch Cleanup		20
	DSM	660	
	DSM Cleanup		30
	Earth Flyby Targeting/Bias		50
	Jupiter Flyby Targeting		30
	Phase Subtotal	660	130
Capture & Probe Release	UOI targeting		10
	UOI	1,087	
	UOI Cleanup		45
	Probe Targeting	15	
	Probe Separation		
	Orbiter Divert	29	
	Orbiter/Probe Statistical		5
	PRM	171	
	PRM Cleanup		20
	Phase Subtotal	1,302	80

Tour	Miranda Targeting	0	
	Miranda Statistical		6
	Ariel Targeting	17	
	Ariel Statistical		6
	Umbriel Targeting	0	
	Umbriel Statistical		6
	Titania Targeting	0	
	Titania Statistical		6
	Oberon Targeting	0	
	Oberon Statistical		6
	Phase Subtotal	17	30
Other	Disposal	216	
	Unallocated Margin	148	
	ACS		125
	Subtotal	364	125
Mission Total		2343	365
		2708	

[7]

Scientific Probe Technology

Scientific payload / in-situ probe:

- <https://pmc.ncbi.nlm.nih.gov/articles/PMC7099645/#S5>
(reviewing previous design concepts for an Ice Giant probe)
 - Note technical challenges mentioned in the article for next slide
- Article on in situ probes ([link](#))
- Atmosphere (chemical composition)
- Temperature
- Radar / comms

Exoplanet Research

Contributors: Cynthia
Presenting: Cynthia



Over half of currently discovered exoplanets are categorized as sub-neptunes [2]

Mini-Neptunes Sub-Neptunes



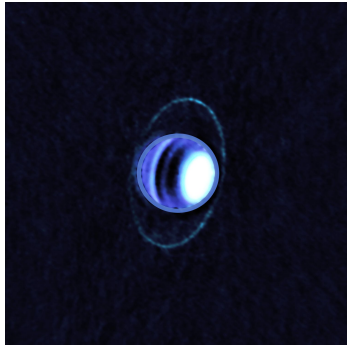
Uranus' Atmosphere

Contributors: Cynthia
Presenting: Cynthia



Coldest atmosphere in the solar system, most transparent

- Atmosphere is particularly hard to detect! Could give info on light curves of exosolar systems w/ sub-Neptunes [14]
- Avg temp: -320deg F ([link](#)) [14]



Composite image of Uranus's atmosphere and rings at radio wavelengths, taken with the Atacama Large Millimeter/submillimeter Array (ALMA) in December 2017. The image shows thermal emission, or heat, from the rings of Uranus for the first time, enabling scientists to determine their temperature is a frigid 77 K (-320 F). Dark bands in Uranus's atmosphere at these wavelengths show the presence of radiolight-absorbing molecules, in particular hydrogen sulfide (H₂S) gas, whereas bright regions like the north polar spot contain very few of these molecules [5]

- Little info about moons
- Really chemically interesting (on same axis)
 - Info about planet formation
 - Might have oceans (we don't know)
- Could inform us about the makeup of exomoons and the chemical signatures to look for
- <https://www.space.com/22201-uranus-moons.html>

Why probe and not just JWST

Contributors: Tamara, Cynthia
Presenting: Tamara

- JWST cannot
- Atmospheric composition (cannot gain all necessary information from external measurements)
 - How does [atmospheric circulation](#) function, from interior to [thermosphere](#), in an ice giant?
(https://en.wikipedia.org/wiki/Uranus_Orbiter_and_Probe)

Why Uranus and not Neptune

Contributors: Cynthia
Presenting: Cynthia

[19]

- Axial tilt is really interesting
 - Magnetic field
- Atmospheric condition
- Flexible launch dates
- Voyager 2 – solar storm

Uranus Orbiter and Probe

- End-to-end viable mission concept on currently available launch vehicle
- Flexible launch dates starting in 2031 through 2038+
- No new technologies required
- Low-Medium risk (only large mission TRACed to receive this)

Neptune Odyssey

- Lacks demonstrated trajectory and launch date within the decade on currently available launch vehicle
- Uncertainties in power requirements and possible need for solar electric propulsion if neither SLS nor Jupiter gravity assist are available
- Accommodation on current launch vehicles unclear (faring size)